

### FLEXURE OF THE LITHOSPHERE (T&S, CH3)

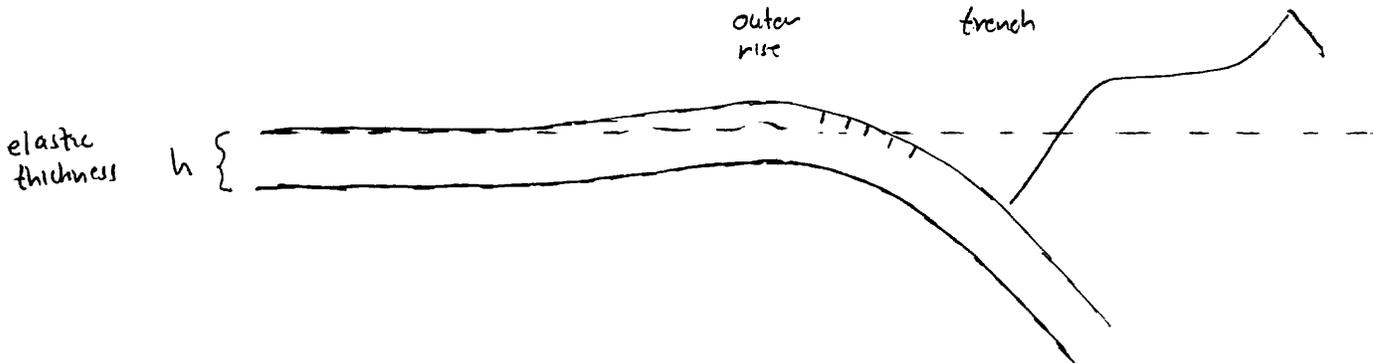
(Veeats, Isostasy and Flexure of the Lithosphere, Cambridge 2001)

How does the outer shell of the earth respond to vertical and end loads?

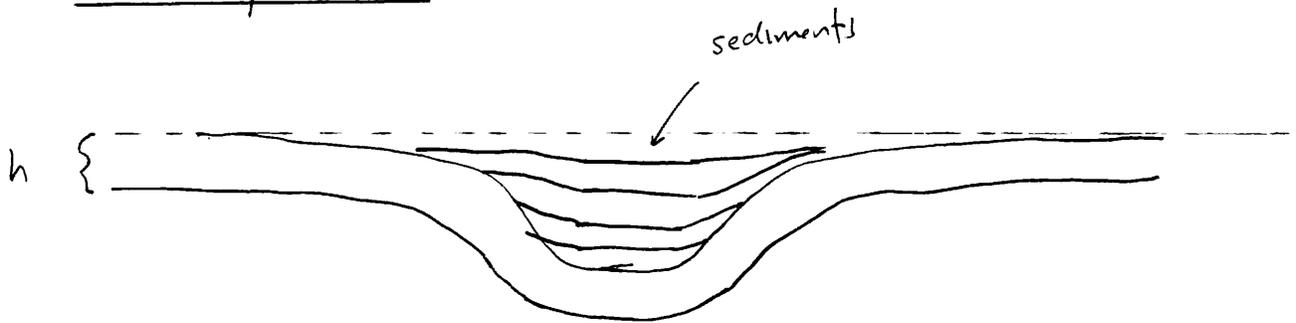
What is the long-term thickness of the lithosphere?

What is the largest load that can be supported by the lithosphere?

#### ocean trench



#### sedimentary basin





last class

principal stress and strain       $\lambda, \mu$

$$\begin{pmatrix} \sigma_1 \\ \sigma_2 \\ \sigma_3 \end{pmatrix} = \begin{pmatrix} \lambda + \mu & \lambda & \lambda \\ \lambda & \lambda + \mu & \lambda \\ \lambda & \lambda & \lambda + \mu \end{pmatrix} \begin{pmatrix} \epsilon_1 \\ \epsilon_2 \\ \epsilon_3 \end{pmatrix}$$

invert  $\Downarrow$

$$\begin{pmatrix} \epsilon_1 \\ \epsilon_2 \\ \epsilon_3 \end{pmatrix} = \frac{1}{E} \begin{pmatrix} 1 & -\nu & -\nu \\ -\nu & 1 & -\nu \\ -\nu & -\nu & 1 \end{pmatrix} \begin{pmatrix} \sigma_1 \\ \sigma_2 \\ \sigma_3 \end{pmatrix} \quad \nu = \frac{\lambda}{2(\lambda + \mu)}$$

$$E = \frac{\mu(3\lambda + 2\mu)}{\lambda + \mu}$$

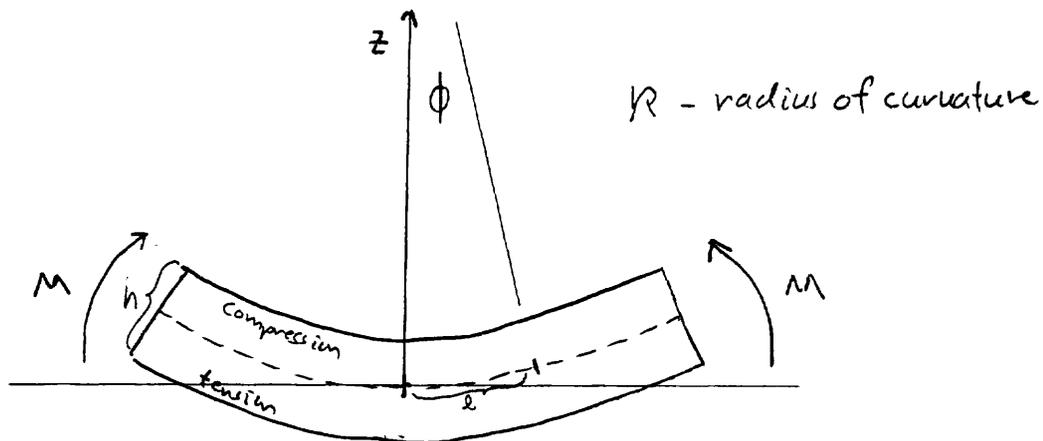
align principal stress/strain axes with x y z

plane stress       $\sigma_{zz} = 0$

save

$$\left\{ \begin{aligned} \epsilon_{xx} &= \frac{1}{E} (\sigma_{xx} - \nu \sigma_{yy}) \\ \epsilon_{yy} &= \frac{1}{E} (\sigma_{yy} - \nu \sigma_{xx}) \\ \epsilon_{zz} &= -\frac{\nu}{E} (\sigma_{xx} + \sigma_{yy}) \end{aligned} \right.$$

moment vs. curvature (T&S p 113-115)



assume 2-dimensional bending of plate

assume thin plate so  $\sigma_{zz} = 0$  - plane stress approximation

assume small displacement

$$M = \int_{-\frac{h}{2}}^{\frac{h}{2}} \sigma_{xx} z \, dz$$

moment per unit length in  
y-direction

$$\epsilon_{xx} = \frac{1}{E} (\sigma_{xx} - \nu \sigma_{yy})$$

$$\epsilon_{yy} = \frac{1}{E} (\sigma_{yy} - \nu \sigma_{xx})$$

$$\epsilon_{yy} = 0 \Rightarrow \sigma_{yy} = \nu \sigma_{xx}$$

$$\epsilon_{xx} = \frac{1-\nu^2}{E} \sigma_{xx}$$

$$M = \frac{E}{(1-\nu^2)} \int_{-\frac{h}{2}}^{\frac{h}{2}} \epsilon_{xx} z \, dz$$

③

$$\epsilon_{xx} = \frac{-\Delta l}{l}$$

$$l(0) = R \phi$$

$$l(z) = (R-z) \phi$$

$$\Delta l(z) = -z \phi$$

$$\epsilon_{xx} = \frac{z}{R} = -z \frac{d^2 w}{dx^2}$$

$$M = \frac{-E}{(1-\nu^2)} \frac{d^2 w}{dx^2} \int_{-\frac{h}{2}}^{\frac{h}{2}} z^2 dz \quad \frac{z^3}{3} \Big|_{-\frac{h}{2}}^{\frac{h}{2}} = \frac{h^3}{12}$$

$$M = \frac{-E h^3}{12(1-\nu^2)} \frac{d^2 w}{dx^2}$$

moment  
(force)

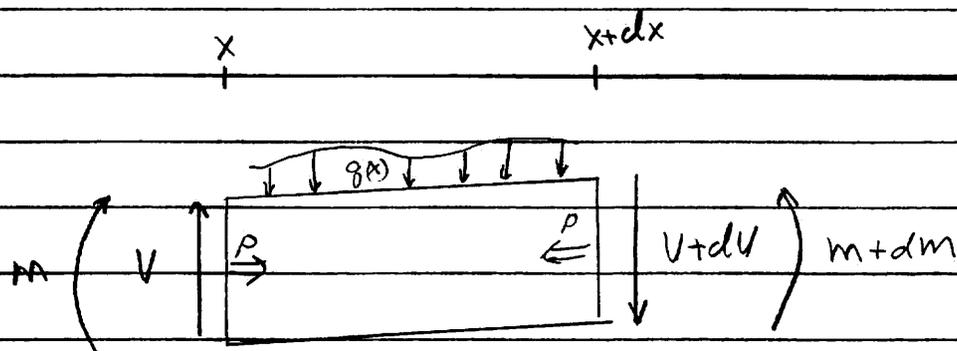
flexural  
rigidity  
(force x  
length)

curvature  
length<sup>-1</sup>

All of the above problems can be approximated by flexure of a thin elastic plate in response to various types of loads.

## 2-D BENDING OR FLEXURE OF PLATES (T&S p. 113-115)

Deflection of a plate  $w(x)$  can be determined by requiring equilibrium under all forces and torques.



$w(x)$  - deflection of plate (length)

$q(x)$  - downward (force/area)

$V$  - shear (force/length)

$m$  - bending moment/length (force length/length = force)

vertical force balance:

$$q(x) dx + dV = 0 \quad \text{or} \quad \frac{dV}{dx} = -q$$

moment balance

$$dm - p dx = V dx \quad \text{or} \quad \frac{dm}{dx} = V + p \frac{dw}{dx}$$

take derivative w.r.t.  $x$

(3)

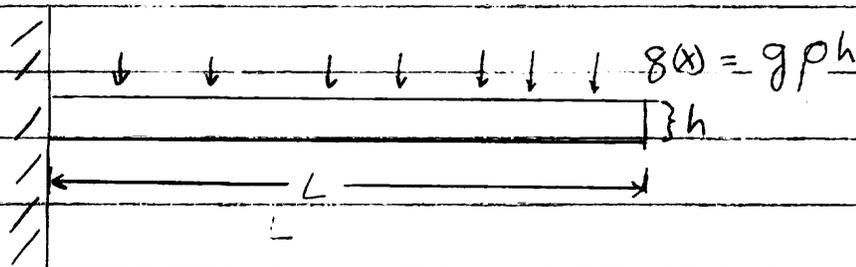
$$\frac{d^2 m}{dx^2} = \frac{dU}{dx} + P \frac{d^2 w}{dx^2} \quad \text{but } \frac{dU}{dx} = -g$$

$$\frac{d^2 m}{dx^2} = -g(x) + P \frac{d^2 w}{dx^2}$$

$$\text{but } m = -D \frac{d^2 w}{dx^2}$$

$$D \frac{d^4 w}{dx^4} + P \frac{d^2 w}{dx^2} = g(x)$$

Example - measuring flexural rigidity of a ruler



$$D \frac{d^4 w}{dx^4} = gph$$

$$w(0) = 0$$

$$\frac{dw}{dx}(0) = 0$$

$$\frac{d^2 w}{dx^2}(L) = 0 \quad \text{no moment}$$

$$\frac{d^3 w}{dx^3}(L) = 0 \quad \text{no load at } x=L$$

$$\frac{d^3w}{dx^3} = \frac{\rho g h}{D} (x + c_1) \quad c_1 = -L$$

$$\frac{d^2w}{dx^2} = \frac{\rho g h}{D} \left( \frac{x^2}{2} - Lx + c_2 \right) \quad c_2 = \frac{L^2}{2}$$

$$\frac{dw}{dx} = \frac{\rho g h}{D} \left( \frac{x^3}{6} - \frac{Lx^2}{2} + \frac{L^2x}{2} + c_3 \right) \quad c_3 = 0$$

$$w = \frac{\rho g h}{D} \left( \frac{x^4}{24} - \frac{Lx^3}{6} + \frac{L^2x^2}{4} \right) \quad c_4 = 0$$

extend ruler over edge of table and measure  $w(L)$

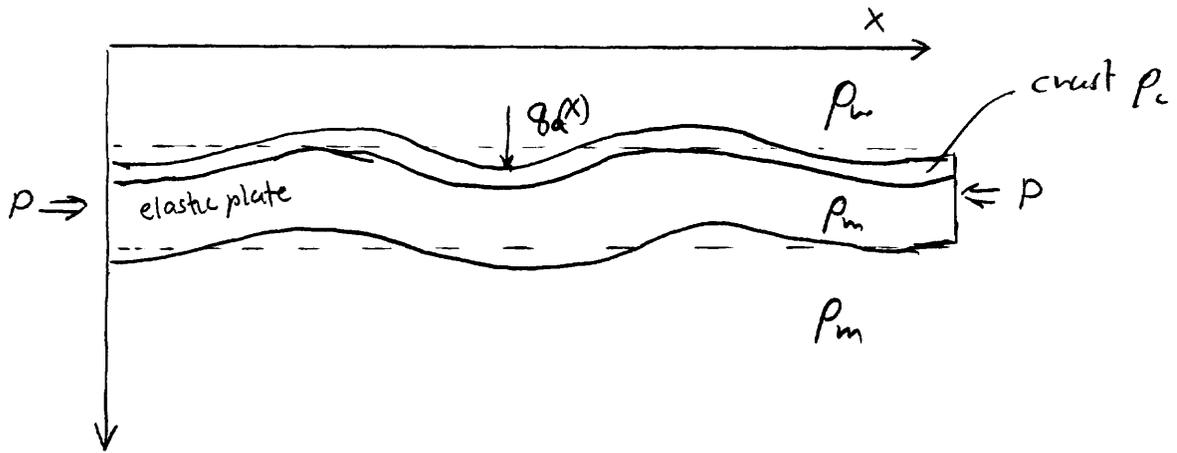
$$w(L) = \frac{\rho g h}{D} \frac{L^4}{8} \quad \text{but } D = \frac{E h^3}{12(1-\nu^2)}$$

$$\frac{E h^3}{12(1-\nu^2)} = \frac{\rho g h L^4}{w(L) 8}$$

$$E = \frac{3(1-\nu^2) \rho g L^4}{w(L) h^2}$$

(7)

stability of an elastic plate under an end load



$$D \frac{d^4 w}{dx^4} + P \frac{d^2 w}{dx^2} + g(\rho_m - \rho_w) w = q_a(x)$$

take fourier transform

$$(2\pi k)^4 D w(k) - (2\pi k)^2 P w(k) + (\rho_m - \rho_w) g w(k) = 0$$

$$D (2\pi k)^4 - P (2\pi k)^2 + (\rho_m - \rho_w) g = 0 \quad ay^2 + by + c = 0$$

$$(2\pi k)^2 = \frac{P \pm \sqrt{P^2 - 4D(\rho_m - \rho_w)g}}{2D}$$

$$y = \frac{-b \pm \sqrt{b^2 - 4ac}}{2a}$$

$$P > [4Dg(\rho_m - \rho_w)]^{\frac{1}{2}}$$

wavenumber must be real so  $P > [4Dg(\rho_m - \rho_w)]^{\frac{1}{2}}$

Critical wavenumber  $P_c = [4Dg(\rho_m - \rho_w)]^{\frac{1}{2}}$   $\sigma_c = \frac{P_c}{h}$

critical wavelength  $(2\pi k_c)^2 = \frac{[4Dg(\rho_m - \rho_w)]^{\frac{1}{2}}}{2D}$

$$\lambda_c = \frac{1}{k_c} = 2\pi \left( \frac{D}{g(\rho_m - \rho_w)} \right)^{\frac{1}{4}}$$

later we'll see this is the flexural wavelength

Example  $h = 50 \text{ km}$   $E = 10^{11} \text{ Pa}$   $\nu = .25$   
 $\rho_m = 3300$   $\rho_w = 1025$

$\sigma = P_c/h = 6.4 \text{ GPa}$  impossible!  
 $\lambda_c = 513 \text{ km}$

